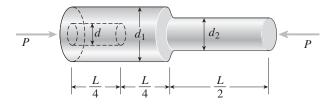
Problem 2.3-8 A bar *ABC* of length *L* consists of two parts of equal lengths but different diameters (see figure). Segment *AB* has diameter $d_1 = 100$ mm and segment *BC* has diameter $d_2 = 60$ mm. Both segments have length L/2 = 0.6 m. A longitudinal hole of diameter *d* is drilled through segment *AB* for one-half of its length (distance L/4 = 0.3 m). The bar is made of plastic having modulus of elasticity E = 4.0 GPa. Compressive loads P = 110 kN act at the ends of the bar.

If the shortening of the bar is limited to 8.0 mm, what is the maximum allowable diameter d_{max} of the hole?

Solution 2.3-8 Bar with a hole



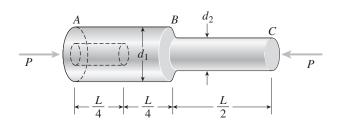
d = diameter of hole

Shortening δ of the bar

$$\delta = \sum \frac{N_i L_i}{E_i A_i} = \frac{P}{E} \sum \frac{L_i}{A_i}$$
$$= \frac{P}{E} \left[\frac{L/4}{\frac{\pi}{4}(d_1^2 - d^2)} + \frac{L/4}{\frac{\pi}{4}d_1^2} + \frac{L/2}{\frac{\pi}{4}d_2^2} \right]$$
$$= \frac{PL}{\pi E} \left(\frac{1}{d_1^2 - d^2} + \frac{1}{d_1^2} + \frac{2}{d_2^2} \right)$$
(Eq. 1)

NUMERICAL VALUES (DATA):

 δ = maximum allowable shortening of the bar = 8.0 mm



P = 110 kN L = 1.2 m E = 4.0 GPa

 $d_1 = 100 \text{ mm}$

- $d_{\text{max}} =$ maximum allowable diameter of the hole $d_2 = 60 \text{ mm}$
- Substitute numerical values into Eq. (1) for δ and solve for $d = d_{max}$:

UNITS: Newtons and meters

$$0.008 = \frac{(110,000)(1.2)}{\pi(4.0 \times 10^9)} \\ \times \left[\frac{1}{(0.1)^2 - d^2} + \frac{1}{(0.1)^2} + \frac{2}{(0.06)^2}\right] \\ 761.598 = \frac{1}{0.01 - d^2} + \frac{1}{0.01} + \frac{2}{0.0036} \\ \frac{1}{0.01 - d^2} = 761.598 - 100 - 555.556 = 106.042 \\ d^2 = 569.81 \times 10^{-6} \text{ m}^2 \\ d = 0.02387 \text{ m}$$

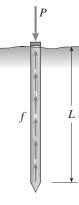
$$d_{\text{max}} = 23.9 \text{ mm} \longleftarrow$$

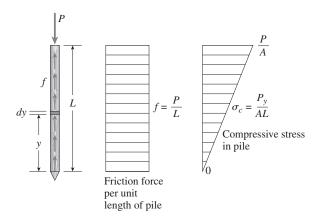
Problem 2.3-9 A wood pile, driven into the earth, supports a load P entirely by friction along its sides (see figure). The friction force f per unit length of pile is assumed to be uniformly distributed over the surface of the pile. The pile has length L, cross-sectional area A, and modulus of elasticity E.

(a) Derive a formula for the shortening δ of the pile in terms of *P*, *L*, *E*, and *A*.

.....

(b) Draw a diagram showing how the compressive stress σ_c varies throughout the length of the pile.





Solution 2.3-9 Wood pile with friction

FROM FREE-BODY DIAGRAM OF PILE:

$$\Sigma F_{\text{vert}} = 0$$
 \uparrow_+ $\downarrow^ fL - P = 0$ $f = \frac{P}{L}$ (Eq. 1)

(a) Shortening δ of pile:

At distance *y* from the base:

$$N(y) = \text{axial force} \qquad N(y) = fy \qquad (Eq. 2)$$

$$d\delta = \frac{N(y) \, dy}{EA} = \frac{fy \, dy}{EA}$$

$$\delta = \int_0^L d\delta = \frac{f}{EA} \int_0^L y \, dy = \frac{fL^2}{2EA} = \frac{PL}{2EA}$$

$$\delta = \frac{PL}{2EA} \longleftarrow$$

(b) Compressive stress σ_c in pile

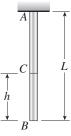
$$\sigma_c = \frac{N(y)}{A} = \frac{fy}{A} = \frac{Py}{AL} \longleftarrow$$

At the base $(y = 0): \sigma_c = 0$
At the top $(y = L): \sigma_c = \frac{P}{A}$

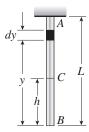
See the diagram above.

Problem 2.3-10 A prismatic bar *AB* of length *L*, cross-sectional area *A*, modulus of elasticity *E*, and weight *W* hangs vertically under its own weight (see figure).

- (a) Derive a formula for the downward displacement δ_C of point *C*, located at distance *h* from the lower end of the bar.
- (b) What is the elongation δ_B of the entire bar?
- (c) What is the ratio β of the elongation of the upper half of the bar to the elongation of the lower half of the bar?



Solution 2.3-10 Prismatic bar hanging vertically



W = Weight of bar(a) DOWNWARD DISPLACEMENT δ_C

Consider an element at distance *y* from the lower end.

$$N(y) = \frac{Wy}{L} \qquad d\delta = \frac{N(y)dy}{EA} = \frac{Wydy}{EAL}$$
$$= \frac{W}{2EAL}(L^2 - h^2)$$
$$\delta_C = \frac{W}{2EAL}(L^2 - h^2) \longleftarrow$$

(b) Elongation of Bar
$$(h = 0)$$

$$\delta_B = \frac{WL}{2EA} \longleftarrow$$

(c) RATIO OF ELONGATIONS

Elongation of upper half of bar $\left(h = \frac{L}{2}\right)$: $\delta_{upper} = \frac{3WL}{2}$

$$upper = \frac{3WL}{8EA}$$

Elongation of lower half of bar:

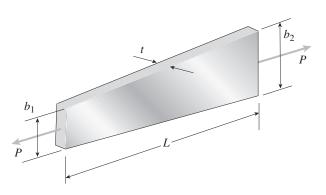
$$\delta_{\text{lower}} = \delta_B - \delta_{\text{upper}} = \frac{WL}{2EA} - \frac{3WL}{8EA} = \frac{WL}{8EA}$$
$$\beta = \frac{\delta_{\text{upper}}}{\delta_{\text{lower}}} = \frac{3/8}{1/8} = 3 \longleftarrow$$

Problem 2.3-11 A flat bar of rectangular cross section, length L, and constant thickness t is subjected to tension by forces P (see figure). The width of the bar varies linearly from b_1 at the smaller end to b_2 at the larger end. Assume that the angle of taper is small.

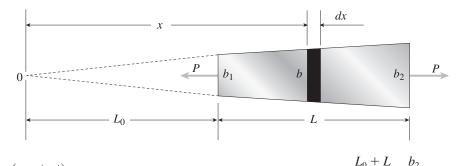
(a) Derive the following formula for the elongation of the bar:

$$\delta = \frac{PL}{Et(b_2 - b_1)} \ln \frac{b_2}{b_1}$$

(b) Calculate the elongation, assuming L = 5 ft, t = 1.0 in., P = 25 k, $b_1 = 4.0$ in., $b_2 = 6.0$ in., and $E = 30 \times 10^6$ psi.



Solution 2.3-11 Tapered bar (rectangular cross section)



t =thickness (constant)

(a) ELONGATION OF THE BAR

$$b = b_1 \left(\frac{x}{L_0}\right) \quad b_2 = b_1 \left(\frac{L_0 + L}{L_0}\right) \tag{Eq. 1}$$
$$A(x) = bt = b_1 t \left(\frac{x}{L_0}\right)$$

From Eq. (1):
$$\frac{-b}{L_0} = \frac{b}{b_1}$$
 (Eq. 3)

Solve Eq. (3) for
$$L_0: L_0 = L\left(\frac{b_1}{b_2 - b_1}\right)$$
 (Eq. 4)

Substitute Eqs. (3) and (4) into Eq. (2):

$$\delta = \frac{PL}{Et(b_2 - b_1)} \ln \frac{b_2}{b_1} \longleftarrow$$
(Eq. 5)

(b) SUBSTITUTE NUMERICAL VALUES:

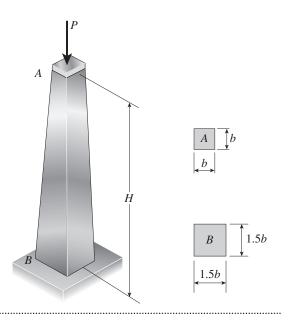
$$L = 5 \text{ ft} = 60 \text{ in.}$$
 $t = 10 \text{ in.}$
 $P = 25 \text{ k}$ $b_1 = 4.0 \text{ in.}$
 $b_2 = 6.0 \text{ in.}$ $E = 30 \times 10^6 \text{ psi}$
From Eq. (5): $\delta = 0.010 \text{ in.}$

$$d\delta = \frac{Pdx}{EA(x)} = \frac{PL_0 dx}{Eb_1 tx}$$
$$\delta = \int_{L_0}^{L_0 + L} d\delta = \frac{PL_0}{Eb_1 t} \int_{L_0}^{L_0 + L} \frac{dx}{x}$$

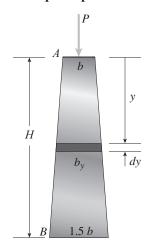
$$= \frac{PL_0}{Eb_1 t} \ln x \Big|_{L_0}^{L_0 + L} = \frac{PL_0}{Eb_1 t} \ln \frac{L_0 + L}{L_0}$$
(Eq. 2)

Problem 2.3-12 A post *AB* supporting equipment in a laboratory is tapered uniformly throughout its height *H* (see figure). The cross sections of the post are square, with dimensions $b \times b$ at the top and $1.5b \times 1.5b$ at the base.

Derive a formula for the shortening δ of the post due to the compressive load *P* acting at the top. (Assume that the angle of taper is small and disregard the weight of the post itself.)



Solution 2.3-12 Tapered post



.....

Square cross sections

b =width at A

1.5b =width at B

 $b_y =$ width at distance y

$$= b + (1.5b - b)\frac{y}{H}$$
$$= \frac{b}{H}(H + 0.5y)$$

 $A_y =$ cross-sectional area at distance y

$$= (b_y)^2 = \frac{b^2}{H^2}(H + 0.5y)^2$$

Shortening of element dy

$$d\delta = \frac{Pdy}{EA_y} = \frac{Pdy}{E\left(\frac{b^2}{H^2}\right)(H+0.5y)^2}$$

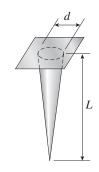
SHORTENING OF ENTIRE POST

$$\delta = \int d\delta = \frac{PH^2}{Eb^2} \int_0^H \frac{dy}{(H+0.5y)^2}$$

From Appendix C: $\int \frac{dx}{(a+bx)^2} = -\frac{1}{b(a+bx)}$
$$\delta = \frac{PH^2}{Eb^2} \left[-\frac{1}{(0.5)(H+0.5y)} \right]_0^H$$
$$= \frac{PH^2}{Eb^2} \left[-\frac{1}{(0.5)(1.5H)} + \frac{1}{0.5H} \right]$$
$$= \frac{2PH}{3Eb^2} \longleftarrow$$

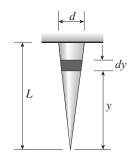
Problem 2.3-13 A long, slender bar in the shape of a right circular cone with length L and base diameter d hangs vertically under the action of its own weight (see figure). The weight of the cone is W and the modulus of elasticity of the material is E.

Derive a formula for the increase δ in the length of the bar due to its own weight. (Assume that the angle of taper of the cone is small.)



.....

Solution 2.3-13 Conical bar hanging vertically



ELEMENT OF BAR

$$\begin{array}{c} \uparrow N_y \\ \downarrow N_y \end{array} \begin{array}{c} \downarrow \\ \uparrow \end{array} dy$$

W=weight of cone

TERMINOLOGY

 N_{y} = axial force acting on element dy

 A_{y} = cross-sectional area at element dy

 $A_B =$ cross-sectional area at base of cone

$$=\frac{\pi d^2}{4}$$

V = volume of cone

$$=\frac{1}{3}A_BL$$

 V_v = volume of cone below element dy

$$=\frac{1}{3}A_y y$$

 W_{y} = weight of cone below element dy

$$= \frac{V_y}{V}(W) = \frac{A_y y V}{A_B L}$$
$$N_y = W_y$$

ELONGATION OF ELEMENT dy

$$d\delta = \frac{N_y \, dy}{E \, A_y} = \frac{Wy \, dy}{E \, A_B L} = \frac{4W}{\pi \, d^2 E L} \, y \, dy$$

ELONGATION OF CONICAL BAR

$$\delta = \int d\delta = \frac{4W}{\pi d^2 EL} \int_0^L y \, dy = \frac{2WL}{\pi d^2 E} \longleftarrow$$

Problem 2.3-14 A bar *ABC* revolves in a horizontal plane about a vertical axis at the midpoint *C* (see figure). The bar, which has length 2*L* and cross-sectional area *A*, revolves at constant angular speed ω . Each half of the bar (*AC* and *BC*) has weight W_1 and supports a weight W_2 at its end.

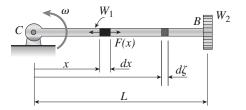
Derive the following formula for the elongation of one-half of the bar (that is, the elongation of either *AC* or *BC*):

$$\delta = \frac{L^2 \omega^2}{3gEA} \left(W_1 + 3W_2 \right)$$

in which *E* is the modulus of elasticity of the material of the bar and *g* is the acceleration of gravity.

.....

Solution 2.3-14 Rotating bar



 ω = angular speed

A = cross-sectional area

E =modulus of elasticity

g =acceleration of gravity

F(x) = axial force in bar at distance x from point C

Consider an element of length dx at distance x from point C.

To find the force F(x) acting on this element, we must find the inertia force of the part of the bar from distance *x* to distance *L*, plus the inertia force of the weight W_2 .

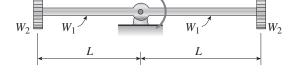
Since the inertia force varies with distance from point *C*, we now must consider an element of length $d\xi$ at distance ξ , where ξ varies from *x* to *L*.

Mass of element $d\xi = \frac{d\xi}{L} \left(\frac{W_1}{g} \right)$

Acceleration of element = $\xi \omega^2$

Centrifugal force produced by element

= (mass)(acceleration) =
$$\frac{W_1\omega^2}{gL} \xi d\xi$$



85

Centrifugal force produced by weight W_2

$$= \left(\frac{W_2}{g}\right)(L\omega^2)$$

AXIAL FORCE F(x)

$$F(x) = \int_{\xi=x}^{\xi=L} \frac{W_1 \omega^2}{gL} \,\xi d\xi + \frac{W_2 L \omega^2}{g}$$
$$= \frac{W_1 \omega^2}{2gL} (L^2 - x^2) + \frac{W_2 L \omega^2}{g}$$

ELONGATION OF BAR BC

$$\delta = \int_0^L \frac{F(x) dx}{EA}$$

$$= \int_0^L \frac{W_1 \omega^2}{2gLEA} (L^2 - x^2) dx + \int_0^L \frac{W_2 L \omega^2 dx}{gEA}$$

$$= \frac{W_1 \omega^2}{2gLEA} \left[\int_0^L L^2 dx - \int_0^L x^2 dx \right] + \frac{W_2 L \omega^2}{gEA} \int_0^L dx$$

$$= \frac{W_1 L^2 \omega^2}{3gEA} + \frac{W_2 L^2 \omega^2}{gEA}$$

$$= \frac{L^2 \omega^2}{3gEA} (W_1 + 3W_2) \longleftarrow$$

Problem 2.3-15 The main cables of a suspension bridge [see part (a) of the figure] follow a curve that is nearly parabolic because the primary load on the cables is the weight of the bridge deck, which is uniform in intensity along the horizontal. Therefore, let us represent the central region AOB of one of the main cables [see part (b) of the figure] as a parabolic cable supported at points A and B and carrying a uniform load of intensity q along the horizontal. The span of the cable is L, the sag is h, the axial rigidity is EA, and the origin of coordinates is at midspan.

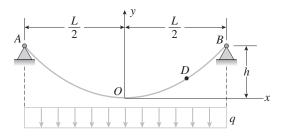
(a) Derive the following formula for the elongation of cable *AOB* shown in part (b) of the figure:

$$\delta = \frac{qL^3}{8hEA} \left(1 + \frac{16h^2}{3L^2}\right)$$

(b) Calculate the elongation δ of the central span of one of the main cables of the Golden Gate Bridge, for which the dimensions and properties are L = 4200 ft, h = 470 ft, q = 12,700 lb/ft, and E = 28,800,000 psi. The cable consists of 27,572 parallel wires of diameter 0.196 in.

Hint: Determine the tensile force *T* at any point in the cable from a free-body diagram of part of the cable; then determine the elongation of an element of the cable of length *ds*; finally, integrate along the curve of the cable to obtain an equation for the elongation δ .

Solution 2.3-15 Cable of a suspension bridge



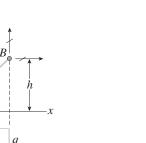
D

H O

Equation of parabolic curve:

$$y = \frac{4hx^2}{L^2}$$
$$\frac{dy}{dx} = \frac{8hx}{L^2}$$

FREE-BODY DIAGRAM OF HALF OF CABLE



 $-Hh + \frac{qL}{2}\left(\frac{L}{4}\right) = 0$ $H = \frac{qL^2}{8h}$

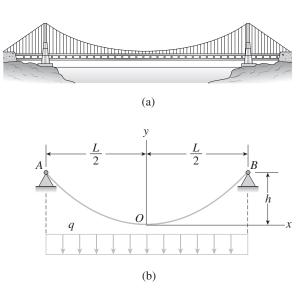
 $\Sigma F_{\text{horizontal}} = 0$

 $\Sigma M_B = 0$ for $M_B = 0$

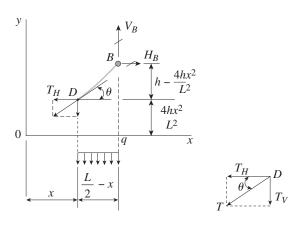
$$H_B = H = \frac{qL^2}{8h}$$
(Eq. 1)

$$\Sigma F_{\text{vertical}} = 0$$

$$V_B = \frac{qL}{2}$$
(Eq. 2)



Free-body diagram of segment DB of cable



$$\Sigma F_{\text{horiz}} = 0 \qquad T_H = H_B$$

$$= \frac{qL^2}{8h} \qquad (\text{Eq. 3})$$

$$\Sigma F_{\text{vert}} = 0 \qquad V_B - T_V - q\left(\frac{L}{2} - x\right) = 0$$

$$T_V = V_B - q\left(\frac{L}{2} - x\right) = \frac{qL}{2} - \frac{qL}{2} + qx$$

$$= qx \qquad (\text{Eq. 4})$$

Tensile force T in cable

$$T = \sqrt{T_H^2 + T_V^2} = \sqrt{\left(\frac{qL^2}{8h}\right)^2 + (qx)^2}$$
$$= \frac{qL^2}{8h}\sqrt{1 + \frac{64h^2x^2}{L^4}}$$
(Eq. 5)

Elongation $d\delta$ of an element of length ds

$$ds = \frac{Tds}{EA}$$

$$ds = \sqrt{(dx)^2 + (dy)^2} = dx \sqrt{1 + \left(\frac{dy}{dx}\right)^2}$$

$$= dx \sqrt{1 + \left(\frac{8hx}{L^2}\right)^2}$$

$$= dx \sqrt{1 + \frac{64h^2x^2}{L^4}}$$
(Eq. 6)

(a) Elongation δ of cable AOB

$$\delta = \int d\delta = \int \frac{T \, ds}{EA}$$

Substitute for *T* from Eq. (5) and for ds from Eq. (6):

$$\delta = \frac{1}{EA} \int \frac{qL^2}{8h} \left(1 + \frac{64h^2 x^2}{L^4} \right) dx$$

For both halves of cable:

$$\delta = \frac{2}{EA} \int_{0}^{L/2} \frac{qL^2}{8h} \left(1 + \frac{64h^2x^2}{L^4}\right) dx$$
$$\delta = \frac{qL^3}{8hEA} \left(1 + \frac{16h^2}{3L^2}\right) \longleftarrow$$
(Eq. 7)

(b) GOLDEN GATE BRIDGE CABLE

$$L = 4200 \text{ ft}$$
 $h = 470 \text{ ft}$
 $q = 12,700 \text{ lb/ft}$ $E = 28,800,000 \text{ psi}$

27,572 wires of diameter d = 0.196 in.

$$A = (27,572) \left(\frac{\pi}{4}\right) (0.196 \text{ in.})^2 = 831.90 \text{ in.}^2$$

Substitute into Eq. (7):

$$\delta = 133.7$$
 in $= 11.14$ ft \leftarrow

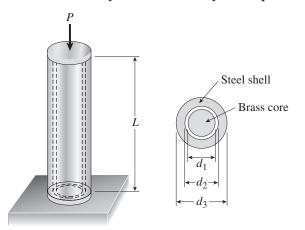
Statically Indeterminate Structures

Problem 2.4-1 The assembly shown in the figure consists of a brass core (diameter $d_1 = 0.25$ in.) surrounded by a steel shell (inner diameter $d_2 = 0.28$ in., outer diameter $d_3 = 0.35$ in.). A load *P* compresses the core and shell, which have length L = 4.0 in. The moduli of elasticity of the brass and steel are $E_b = 15 \times 10^6$ psi and $E_s = 30 \times 10^6$ psi, respectively.

- (a) What load *P* will compress the assembly by 0.003 in.?
- (b) If the allowable stress in the steel is 22 ksi and the allowable stress in the brass is 16 ksi, what is the allowable compressive load $P_{\rm allow}$? (*Suggestion:* Use the equations derived in Example 2-5.)



Solution 2.4-1 Cylindrical assembly in compression



$$d_1 = 0.25 \text{ in.} \qquad E_b = 15 \times 10^6 \text{ psi}$$

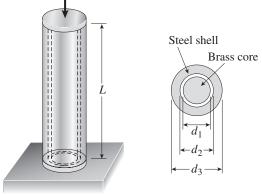
$$d_2 = 0.28 \text{ in.} \qquad E_s = 30 \times 10^6 \text{ psi}$$

$$d_3 = 0.35 \text{ in.} \qquad A_s = \frac{\pi}{4} (d_3^2 - d_2^2) = 0.03464 \text{ in.}^2$$

$$L = 4.0 \text{ in.} \qquad A_b = \frac{\pi}{4} d_1^2 = 0.04909 \text{ in.}^2$$

(a) DECREASE IN LENGTH ($\delta = 0.003$ in.) Use Eq. (2-13) of Example 2-5.

$$\delta = \frac{PL}{E_s A_s + E_b A_b} \quad \text{or}$$
$$P = (E_s A_s + E_b A_b) \left(\frac{\delta}{L}\right)$$



Substitute numerical values:

$$E_{s} A_{s} + E_{b} A_{b} = (30 \times 10^{6} \text{ psi})(0.03464 \text{ in.}^{2}) + (15 \times 10^{6} \text{ psi})(0.04909 \text{ in.}^{2}) = 1.776 \times 10^{6} \text{ lb}$$
$$P = (1.776 \times 10^{6} \text{ lb}) \left(\frac{0.003 \text{ in.}}{4.0 \text{ in.}}\right) = 1330 \text{ lb} \longleftarrow$$
(b) Allowable load $\sigma_{s} = 22 \text{ ksi}$ $\sigma_{b} = 16 \text{ ksi}$

Use Eqs. (2-12a and b) of Example 2-5.

For steel:

$$\sigma_s = \frac{PE_s}{E_s A_s + E_b A_b} \quad P_s = (E_s A_s + E_b A_b) \frac{\sigma_s}{E_s}$$
$$P_s = (1.776 \times 10^6 \text{ lb}) \left(\frac{22 \text{ ksi}}{30 \times 10^6 \text{ psi}}\right) = 1300 \text{ lb}$$

For brass:

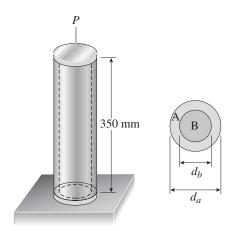
$$\sigma_b = \frac{PE_b}{E_s A_s + E_b A_b} \quad P_s = (E_s A_s + E_b A_b) \frac{\sigma_b}{E_b}$$
$$P_s = (1.776 \times 10^6 \text{ lb}) \left(\frac{16 \text{ ksi}}{15 \times 10^6 \text{ psi}}\right) = 1890 \text{ lb}$$
Steel governs. $P_{\text{allow}} = 1300 \text{ lb} \longleftarrow$

Problem 2.4-2 A cylindrical assembly consisting of a brass core and an aluminum collar is compressed by a load P (see figure). The length of the aluminum collar and brass core is 350 mm, the diameter of the core is 25 mm, and the outside diameter of the collar is 40 mm. Also, the moduli of elasticity of the aluminum and brass are 72 GPa and 100 GPa, respectively.

- (a) If the length of the assembly decreases by 0.1% when the load *P* is applied, what is the magnitude of the load?
- (b) What is the maximum permissible load P_{max} if the allowable stresses in the aluminum and brass are 80 MPa and 120 MPa, respectively? (Suggestion: Use the equations derived in Example 2-5.)



.....



A = aluminum

B = brass

L = 350 mm

 $d_{a} = 40 \text{ mm}$

 $d_{h} = 25 \text{ mm}$

$$A_a = \frac{\pi}{4} (d_a^2 - d_b^2)$$

 $=765.8 \text{ mm}^2$

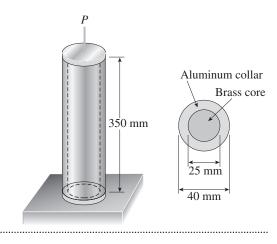
$$E_a = 72 \text{ GPa} \quad E_b = 100 \text{ GPa} \quad A_b = \frac{\pi}{4} d_b^2$$

$$= 490.9 \text{ mm}^2$$

(a) DECREASE IN LENGTH

$$(\delta = 0.1\% \text{ of } L = 0.350 \text{ mm})$$

Use Eq. (2-13) of Example 2-5.



$$\delta = \frac{PL}{E_a A_a + E_b A_b} \quad \text{or}$$
$$P = (E_a A_a + E_b A_b) \left(\frac{\delta}{L}\right)$$

....

Substitute numerical values:

. . .

.

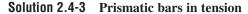
$$E_{a}A_{a} + E_{b}A_{b} = (72 \text{ GPa})(765.8 \text{ mm}^{2}) + (100 \text{ GPa})(490.9 \text{ mm}^{2}) = 55.135 \text{ MN} + 49.090 \text{ MN} = 104.23 \text{ MN}$$
$$P = (104 \cdot 23 \text{ MN}) \left(\frac{0.350 \text{ mm}}{350 \text{ mm}}\right) = 104.2 \text{ kN} \longleftarrow$$
(b) ALLOWABLE LOAD
$$\sigma_{A} = 80 \text{ MPa} \qquad \sigma_{b} = 120 \text{ MPa}$$
Use Eqs. (2-12a and b) of Example 2-5.
For aluminum:
$$\sigma_{a} = \frac{PE_{a}}{E_{a}A_{a} + E_{b}A_{b}} \quad P_{a} = (E_{a}A_{a} + E_{b}A_{b}) \left(\frac{\sigma_{a}}{E_{a}}\right) = 115.8 \text{ kN}$$

For brass:

$$\sigma_b = \frac{PE_b}{E_a A_a + E_b A_b} \quad P_b = (E_a A_a + E_b A_b) \left(\frac{\sigma_b}{E_b}\right)$$
$$P_b = (104.23 \text{ MN}) \left(\frac{120 \text{ MPa}}{100 \text{ GPa}}\right) = 125.1 \text{ kN}$$
Aluminum governs. $P_{\text{max}} = 116 \text{ kN}$

Problem 2.4-3 Three prismatic bars, two of material A and one of material B, transmit a tensile load P (see figure). The two outer bars (material A) are identical. The cross-sectional area of the middle bar (material B) is 50% larger than the cross-sectional area of one of the outer bars. Also, the modulus of elasticity of material A is twice that of material B.

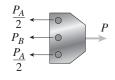
- (a) What fraction of the load *P* is transmitted by the middle bar?
- (b) What is the ratio of the stress in the middle bar to the stress in the outer bars?
- (c) What is the ratio of the strain in the middle bar to the strain in the outer bars?





 $\begin{array}{c} A \\ \hline B \\ \hline O \\ \hline A \end{array} \xrightarrow{O} P$

FREE-BODY DIAGRAM OF END PLATE



EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{horiz}} = 0 \qquad P_A + P_B - P = 0 \tag{1}$$

EQUATION OF COMPATIBILITY

$$\delta_A = \delta_B \tag{2}$$

FORCE-DISPLACEMENT RELATIONS

 A_A = total area of both outer bars

$$\delta_A = \frac{P_A L}{E_A A_A} \quad \delta_B = \frac{P_B L}{E_B A_B} \tag{3}$$

Substitute into Eq. (2):

$$\frac{P_A L}{E_A A_A} = \frac{P_B L}{E_B A_B} \tag{4}$$

SOLUTION OF THE EQUATIONS

Solve simultaneously Eqs. (1) and (4):

$$P_A = \frac{E_A A_A P}{E_A A_A + E_B A_B} \quad P_B = \frac{E_B A_B P}{E_A A_A + E_B A_B} \tag{5}$$

Substitute into Eq. (3):

$$\delta = \delta_A = \delta_B = \frac{PL}{E_A A_A + E_B A_B} \tag{6}$$

STRESSES:

$$\sigma_A = \frac{P_A}{A_A} = \frac{E_A P}{E_A A_A + E_B A_B} \quad \sigma_B = \frac{P_B}{A_B}$$
$$= \frac{E_B P}{E_A A_A + E_B A_B} \tag{7}$$

(a) LOAD IN MIDDLE BAR

$$\frac{P_B}{P} = \frac{E_B A_B}{E_A A_A + E_B A_B} = \frac{1}{\frac{E_A A_A}{E_B A_B} + 1}$$

Given: $\frac{E_A}{E_B} = 2$ $\frac{A_A}{A_B} = \frac{1+1}{1.5} = \frac{4}{3}$
 $\therefore \frac{P_B}{P} = \frac{1}{\left(\frac{E_A}{E_B}\right) \left(\frac{A_A}{A_B}\right) + 1} = \frac{1}{\frac{8}{3} + 1} = \frac{3}{11} \longleftarrow$

(b) RATIO OF STRESSES

$$\frac{\sigma_B}{\sigma_A} = \frac{E_B}{E_A} = \frac{1}{2} \longleftarrow$$

(c) Ratio of strains

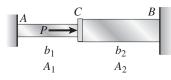
All bars have the same strain

Ratio = $1 \leftarrow --$

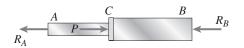
Problem 2.4-4 A bar *ACB* having two different cross-sectional areas A_1 and A_2 is held between rigid supports at *A* and *B* (see figure). A load *P* acts at point *C*, which is distance b_1 from end *A* and distance b_2 from end *B*.

- (a) Obtain formulas for the reactions R_A and R_B at supports A and B, respectively, due to the load P.
- (b) Obtain a formula for the displacement δ_C of point *C*.
- (c) What is the ratio of the stress σ_1 in region AC to the stress σ_2 in region CB?

Solution 2.4-4 Bar with intermediate load



FREE-BODY DIAGRAM



EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{horiz}} = 0 \qquad \qquad R_A + R_B = P \qquad \text{(Eq. 1)}$$

EQUATION OF COMPATIBILITY

 δ_{AC} = elongation of AC

 δ_{CB} = shortening of *CB*

$$\delta_{AC} = \delta_{CB} \tag{Eq. 2}$$

FORCE DISPLACEMENT RELATIONS

$$\delta_{AC} = \frac{R_A b_1}{EA_1} \quad \delta_{CB} = \frac{R_B b_2}{EA_2}$$
(Eqs. 3&4)

(a) SOLUTION OF EQUATIONS

Substitute Eq. (3) and Eq. (4) into Eq. (2):

$$\frac{R_A b_1}{EA_1} = \frac{R_B b_2}{EA_2}$$
(Eq. 5)

Solve Eq. (1) and Eq. (5) simultaneously:

$$R_A = \frac{b_2 A_1 P}{b_1 A_2 + b_2 A_1} \quad R_B = \frac{b_1 A_2 P}{b_1 A_2 + b_2 A_1} \longleftarrow$$

(b) DISPLACEMENT OF POINT C

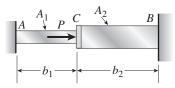
$$\delta_C = \delta_{AC} = \frac{R_A b_1}{EA_1} = \frac{b_1 b_2 P}{E(b_1 A_2 + b_2 A_1)} \longleftarrow$$

(c) RATIO OF STRESSES

$$\sigma_1 = \frac{R_A}{A_1} \text{ (tension)} \quad \sigma_2 = \frac{R_B}{A_2} \text{ (compression)}$$

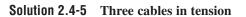
 $\frac{\sigma_1}{\sigma_2} = \frac{b_2}{b_1} \longleftarrow$

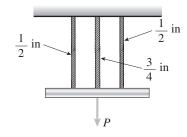
(Note that if $b_1 = b_2$, the stresses are numerically equal regardless of the areas A_1 and A_2 .)



Problem 2.4-5 Three steel cables jointly support a load of 12 k (see figure). The diameter of the middle cable is $\frac{34}{4}$ in. and the diameter of each outer cable is $\frac{14}{2}$ in. The tensions in the cables are adjusted so that each cable carries one-third of the load (i.e., 4 k). Later, the load is increased by 9 k to a total load of 21 k.

- (a) What percent of the total load is now carried by the middle cable?
- (b) What are the stresses σ_M and σ_O in the middle and outer cables, respectively? (*Note:* See Table 2-1 in Section 2.2 for properties of cables.)





AREAS OF CABLES (from Table 2-1)

Middle cable: $A_M = 0.268$ in.²

Outer cables: $A_o = 0.119$ in.²

(for each cable)

FIRST LOADING

$$P_1 = 12 \text{ k} \left(\text{Each cable carries } \frac{P_1}{3} \text{ or } 4 \text{ k.} \right)$$

SECOND LOADING

 $P_2 = 9 \text{ k} \text{ (additional load)}$

EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{vert}} = 0 \qquad \qquad 2P_o + P_M - P_2 = 0 \qquad (1)$$

EQUATION OF COMPATIBILITY

$$\delta_M = \delta_o \tag{2}$$

FORCE-DISPLACEMENT RELATIONS

$$\delta_M = \frac{P_M L}{EA_M} \quad \delta_o = \frac{P_o L}{EA_o} \tag{3, 4}$$

SUBSTITUTE INTO COMPATIBILITY EQUATION:

$$\frac{P_M L}{EA_M} = \frac{P_o L}{EA_o} \quad \frac{P_M}{A_M} = \frac{P_o}{A_o} \tag{5}$$

Solve simultaneously Eqs. (1) and (5):

$$P_{M} = P_{2} \left(\frac{A_{M}}{A_{M} + 2A_{o}} \right) = (9 \text{ k}) \left(\frac{0.268 \text{ in.}^{2}}{0.506 \text{ in.}^{2}} \right)$$
$$= 4.767 \text{ k}$$
$$P_{o} = P_{2} \left(\frac{A_{o}}{A_{M} + 2A_{o}} \right) = (9 \text{ k}) \left(\frac{0.119 \text{ in.}^{2}}{0.506 \text{ in.}^{2}} \right)$$
$$= 2.117 \text{ k}$$

FORCES IN CABLES

Middle cable: Force = 4 k + 4.767 k = 8.767 kOuter cables: Force = 4 k + 2.117 k = 6.117 k(for each cable)

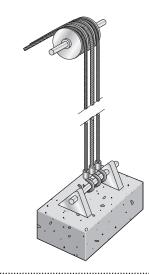
(a) PERCENT OF TOTAL LOAD CARRIED BY MIDDLE CABLE

Percent =
$$\frac{8.767 \text{ k}}{21 \text{ k}}(100\%) = 41.7\%$$
 \longleftarrow

(b) Stresses in Cables ($\sigma = P/A$)

Middle cable:
$$\sigma_M = \frac{8.767 \text{ k}}{0.268 \text{ in.}^2} = 32.7 \text{ ksi} \longleftarrow$$

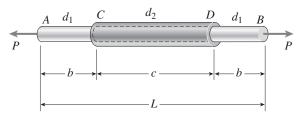
Outer cables: $\sigma_o = \frac{6.117 \text{ k}}{0.119 \text{ in.}^2} = 51.4 \text{ ksi} \longleftarrow$



Problem 2.4-6 A plastic rod *AB* of length L = 0.5 m has a diameter $d_1 = 30$ mm (see figure). A plastic sleeve *CD* of length c = 0.3 m and outer diameter $d_2 = 45$ mm is securely bonded to the rod so that no slippage can occur between the rod and the sleeve. The rod is made of an acrylic with modulus of elasticity $E_1 = 3.1$ GPa and the sleeve is made of a polyamide with $E_2 = 2.5$ GPa.

- (a) Calculate the elongation δ of the rod when it is pulled by axial forces P = 12 kN.
- (b) If the sleeve is extended for the full length of the rod, what is the elongation?
- (c) If the sleeve is removed, what is the elongation?





- $P = 12 \text{ kN} \qquad d_1 = 30 \text{ mm} \qquad b = 100 \text{ mm}$ $L = 500 \text{ mm} \qquad d_2 = 45 \text{ mm} \qquad c = 300 \text{ mm}$ Rod: $E_1 = 3.1 \text{ GPa}$ Sleeve: $E_2 = 2.5 \text{ GPa}$ Rod: $A_1 = \frac{\pi d_1^2}{4} = 706.86 \text{ mm}^2$
- (b) SLEEVE AT FULL LENGTH $S = S = \begin{pmatrix} L \\ - \end{pmatrix} = (0.81815 \text{ mm}) (500 \text{ mm})$

$$\delta = \delta_{CD} \left(\frac{-}{C}\right) = (0.81815 \text{ mm}) \left(\frac{-0.61815}{300 \text{ mm}}\right)$$
$$= 1.36 \text{ mm} \longleftarrow$$

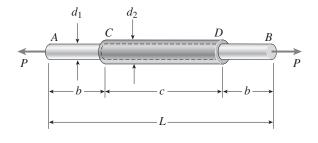
(c) SLEEVE REMOVED

$$\delta = \frac{PL}{E_1 A_1} = 2.74 \text{ mm} \longleftarrow$$

(a) ELONGATION OF ROD Part AC: $\delta_{AC} = \frac{Pb}{E_1A_1} = 0.5476 \text{ mm}$ Part CD: $\delta_{CD} = \frac{Pc}{E_1A_1E_2A_2}$ = 0.81815 mm (From Eq. 2-13 of Example 2-5)

Sleeve: $A_2 = \frac{\pi}{4}(d_2^2 - d_1^2) = 883.57 \text{ mm}^2$

 $E_1A_1 + E_2A_2 = 4.400$ MN



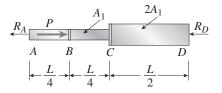
Problem 2.4-7 The axially loaded bar *ABCD* shown in the figure is held between rigid supports. The bar has cross-sectional area A_1 from A to C and $2A_1$ from C to D.

- (a) Derive formulas for the reactions R_A and R_D at the ends of the bar.
- (b) Determine the displacements δ_B and δ_C at points *B* and *C*, respectively.
- (c) Draw a diagram in which the abscissa is the distance from the left-hand support to any point in the bar and the ordinate is the horizontal displacement δ at that point.

Solution 2.4-7 Bar with fixed ends

FREE-BODY DIAGRAM OF BAR

.....



.....

EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{horiz}} = 0$$
 $R_A + R_D = P$ (Eq. 1)

EQUATION OF COMPATIBILITY

$$\delta_{AB} + \delta_{BC} + \delta_{CD} = 0 \tag{Eq. 2}$$

Positive means elongation.

FORCE-DISPLACEMENT EQUATIONS

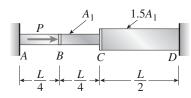
$$\delta_{AB} = \frac{R_A(L/4)}{EA_1}$$
 $\delta_{BC} = \frac{(R_A - P)(L/4)}{EA_1}$ (Eqs. 3, 4)

$$\delta_{CD} = -\frac{R_D(L/2)}{E(2A_1)} \tag{Eq. 5}$$

SOLUTION OF EQUATIONS

Substitute Eqs. (3), (4), and (5) into Eq. (2):

$$\frac{R_A L}{4EA_1} + \frac{(R_A - P)(L)}{4EA_1} - \frac{R_D L}{4EA_1} = 0 \quad (\text{Eq. 6})$$



(a) REAC\TIONS

Solve simultaneously Eqs. (1) and (6):

$$R_A = \frac{2P}{3} \quad R_D = \frac{P}{3} \longleftarrow$$

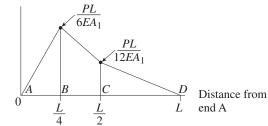
(b) DISPLACEMENTS AT POINTS B and C

$$\delta_B = \delta_{AB} = \frac{R_A L}{4EA_1} = \frac{PL}{6EA_1} \text{ (To the right)} \longleftarrow$$
$$\delta_C = |\delta_{CD}| = \frac{R_D L}{4EA_1}$$

$$= \frac{PL}{12EA_1} \text{ (To the right)} \longleftarrow$$

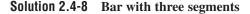
(c) DISPLACEMENT DIAGRAM

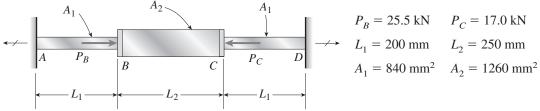
Displacement



Problem 2.4-8 The fixed-end bar *ABCD* consists of three prismatic segments, as shown in the figure. The end segments have crosssectional area $A_1 = 840 \text{ mm}^2$ and length $L_1 = 200 \text{ mm}$. The middle segment has cross-sectional area $A_2 = 1260 \text{ mm}^2$ and length $L_2 = 250$ mm. Loads P_B and P_C are equal to 25.5 kN and 17.0 kN, respectively.

- (a) Determine the reactions R_A and R_D at the fixed supports.
- (b) Determine the compressive axial force F_{BC} in the middle segment of the bar.





FREE-BODY DIAGRAM



EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{horiz}} = 0 \xrightarrow{+} \leftarrow_{-}$$

$$P_{B} + R_{D} - P_{C} - R_{A} = 0 \text{ or}$$

$$R_{A} - R_{D} = P_{B} - P_{C} = 8.5 \text{ kN} \quad (\text{Eq. 1})$$

EQUATION OF COMPATIBILITY

$$\delta_{AD}$$
 = elongation of entire bar
 $\delta_{AD} = \delta_{AB} + \delta_{BC} + \delta_{CD} = 0$ (Eq. 2)

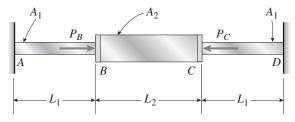
FORCE-DISPLACEMENT RELATIONS

$$\delta_{AB} = \frac{R_A L_1}{E A_1} = \frac{R_A}{E} \left(238.095 \frac{1}{m}\right)$$
 (Eq. 3)

$$\delta_{BC} = \frac{(R_A - P_B)L_2}{EA_2}$$

= $\frac{R_A}{E} \left(198.413 \frac{1}{m} \right) - \frac{P_B}{E} \left(198.413 \frac{1}{m} \right)$ (Eq. 4)

$$\delta_{CD} = \frac{R_D L_1}{E A_1} = \frac{R_D}{E} \left(238.095 \frac{1}{m} \right)$$
(Eq. 5)



SOLUTION OF EQUATIONS

Substitute Eqs. (3), (4), and (5) into Eq. (2):

$$\frac{R_A}{E} \left(238.095 \frac{1}{m} \right) + \frac{R_A}{E} \left(198.413 \frac{1}{m} \right) \\ - \frac{P_B}{E} \left(198.413 \frac{1}{m} \right) + \frac{R_D}{E} \left(238.095 \frac{1}{m} \right) = 0$$

Simplify and substitute $P_B = 25.5$ kN:

$$R_A \left(436.508 \frac{1}{m} \right) + R_D \left(238.095 \frac{1}{m} \right)$$

= 5,059.53 $\frac{\text{kN}}{m}$ (Eq. 6)

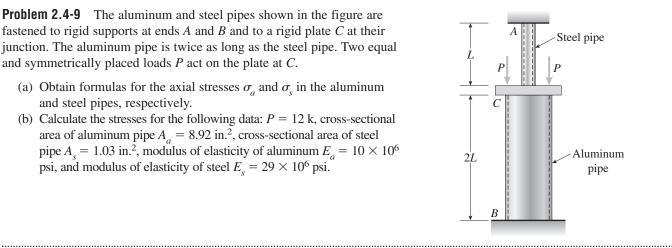
(a) Reactions R_A and R_D Solve simultaneously Eqs. (1) and (6). From (1): $R_D = R_A - 8.5 \text{ kN}$ Substitute into (6) and solve for R_A : $R_A\left(674.603\,\frac{1}{m}\right) = 7083.34\,\frac{\mathrm{kN}}{\mathrm{m}}$ $R_A = 10.5 \text{ kN} \longleftarrow$ $R_D = R_A - 8.5 \text{ kN} = 2.0 \text{ kN} \longleftarrow$

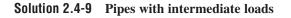
(b) Compressive axial force F_{BC}

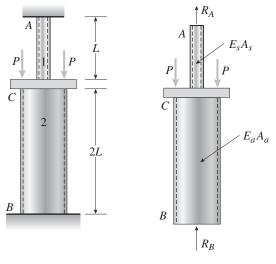
$$F_{BC} = P_B - R_A = P_C - R_D = 15.0 \text{ kN} \longleftarrow$$

Problem 2.4-9 The aluminum and steel pipes shown in the figure are fastened to rigid supports at ends A and B and to a rigid plate C at their junction. The aluminum pipe is twice as long as the steel pipe. Two equal and symmetrically placed loads P act on the plate at C.

- (a) Obtain formulas for the axial stresses σ_a and σ_s in the aluminum and steel pipes, respectively.
- (b) Calculate the stresses for the following data: P = 12 k, cross-sectional area of aluminum pipe $A_a = 8.92$ in.², cross-sectional area of steel pipe $A_s = 1.03$ in.², modulus of elasticity of aluminum $E_a = 10 \times 10^6$ psi, and modulus of elasticity of steel $E_s = 29 \times 10^6$ psi.







1 steel pipe

2 aluminum pipe

EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{vert}} = 0$$
 $R_A + R_B = 2P$ (Eq. 1)

EQUATION OF COMPATIBILITY

$$\delta_{AB} = \delta_{AC} + \delta_{CB} = 0 \tag{Eq. 2}$$

(A positive value of δ means elongation.)

FORCE-DISPLACEMENT RELATIONS

$$\delta_{AC} = \frac{R_A L}{E_s A_s} \quad \delta_{BC} = -\frac{R_B (2L)}{E_a A_a}$$
(Eqs. 3, 4))

SOLUTION OF EQUATIONS

Substitute Eqs. (3) and (4) into Eq. (2):

$$\frac{R_A L}{E_s A_s} - \frac{R_B(2L)}{E_a A_a} = 0$$
 (Eq. 5)

Solve simultaneously Eqs. (1) and (5):

$$R_{A} = \frac{4E_{s}A_{s}P}{E_{a}A_{a} + 2E_{s}A_{s}} \quad R_{B} = \frac{2E_{a}A_{a}P}{E_{a}A_{a} + 2E_{s}A_{s}} \quad (\text{Eqs. 6, 7})$$

(a) AXIAL STRESSES

Aluminum:
$$\sigma_a = \frac{R_B}{A_a} = \frac{2E_a P}{E_a A_a + 2E_s A_s}$$
 (Eq. 8)

(compression)

Steel:
$$\sigma_s = \frac{R_A}{A_s} = \frac{4E_s P}{E_a A_a + 2E_s A_s}$$
 (Eq. 9)
(tension)

(b) NUMERICAL RESULTS

$$P = 12 \text{ k} \qquad A_a = 8.92 \text{ in.}^2 \qquad A_s = 1.03 \text{ in.}^2$$
$$E_a = 10 \times 10^6 \text{ psi} \qquad E_s = 29 \times 10^6 \text{ psi}$$
Substitute into Eqs. (8) and (9):
$$\sigma_a = 1,610 \text{ psi (compression)} \longleftarrow$$
$$\sigma_s = 9,350 \text{ psi (tension)} \longleftarrow$$

Problem 2.4-10 A rigid bar of weight W = 800 N hangs from three equally spaced vertical wires, two of steel and one of aluminum (see figure). The wires also support a load *P* acting at the midpoint of the bar. The diameter of the steel wires is 2 mm, and the diameter of the aluminum wire is 4 mm.

What load P_{allow} can be supported if the allowable stress in the steel wires is 220 MPa and in the aluminum wire is 80 MPa? (Assume $E_s = 210$ GPa and $E_a = 70$ GPa.)



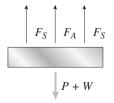
.....

STEEL WIRES $S \quad A \quad S$ W = 800 N P

 $d_s = 2 \text{ mm}$ $\sigma_s = 220 \text{ MPa}$ $E_s = 210 \text{ GPa}$ Aluminum wires

 $d_A = 4 \text{ mm} \qquad \sigma_A = 80 \text{ MPa}$
 $E_A = 70 \text{ GPa}$

FREE-BODY DIAGRAM OF RIGID BAR



EQUATION OF EQUILIBRIUM

$$\begin{split} \Sigma F_{\mathrm{vert}} &= 0\\ 2F_s + F_A - P - W &= 0 \end{split} \tag{Eq. 1}$$

EQUATION OF COMPATIBILITY

$$\delta_s = \delta_A \tag{Eq. 2}$$

FORCE DISPLACEMENT RELATIONS

$$\delta_s = \frac{F_s L}{E_s A_s} \qquad \delta_A = \frac{F_A L}{E_A A_A} \tag{Eqs. 3, 4}$$

SOLUTION OF EQUATIONS

Substitute (3) and (4) into Eq. (2):

$$\frac{F_s L}{E_s A_s} = \frac{F_A L}{E_A A_A} \tag{Eq. 5}$$

Solve simultaneously Eqs. (1) and (5):

$$F_A = (P+W) \left(\frac{E_A A_A}{E_A A_A + 2E_s A_s} \right)$$
(Eq. 6)

$$F_s = (P+W) \left(\frac{E_s A_s}{E_A A_A + 2E_s A_s}\right)$$
(Eq. 7)

STRESSES IN THE WIRES

$$\sigma_A = \frac{F_A}{A_A} = \frac{(P+W)E_A}{E_A A_A + 2E_s A_s}$$
(Eq. 8)

$$\sigma_s = \frac{F_s}{A_s} = \frac{(P+W)E_s}{E_A A_A + 2E_s A_s}$$
(Eq. 9)

Allowable loads (from Eqs. (8) and (9))

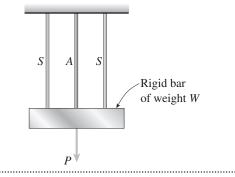
$$P_A = \frac{\sigma_A}{E_A} (E_A A_A + 2E_s A_s) - W$$
 (Eq. 10)

$$P_s = \frac{\sigma_s}{E_s} (E_A A_A + 2E_s A_s) - W$$
 (Eq. 11)

Substitute numerical values into Eqs. (10) and (11):

$$A_s = \frac{\pi}{4} (2 \text{ mm})^2 = 3.1416 \text{ mm}^2$$

 $A_A = \frac{\pi}{4} (4 \text{ mm})^2 = 12.5664 \text{ mm}^2$
 $P_A = 1713 \text{ N}$
 $P_s = 1504 \text{ N}$
Steel governs. $P_{\text{allow}} = 1500 \text{ N} \longleftarrow$

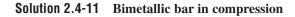


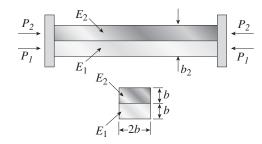
Problem 2.4-11 A *bimetallic* bar (or composite bar) of square cross section with dimensions $2b \times 2b$ is constructed of two different metals having moduli of elasticity E_1 and E_2 (see figure). The two parts of the bar have the same cross-sectional dimensions. The bar is compressed by forces *P* acting through rigid end plates. The line of action of the loads has an eccentricity *e* of such magnitude that each part of the bar is stressed uniformly in compression.

- (a) Determine the axial forces P_1 and P_2 in the two parts of the bar.
- (b) Determine the eccentricity e of the loads.

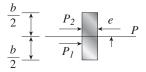
.....

(c) Determine the ratio σ_1/σ_2 of the stresses in the two parts of the bar.





FREE-BODY DIAGRAM (Plate at right-hand end)



EQUATIONS OF EQUILIBRIUM

$$\Sigma F = 0 \qquad P_1 + P_2 = P \tag{Eq. 1}$$

$$\Sigma M = 0 \iff Pe + P_1\left(\frac{b}{2}\right) - P_2\left(\frac{b}{2}\right) = 0$$
 (Eq. 2)

EQUATION OF COMPATIBILITY

$$\delta_2 = \delta_1$$

$$\frac{P_2 L}{E_2 A} = \frac{P_1 L}{E_1 A} \quad \text{or} \quad \frac{P_2}{E_2} = \frac{P_1}{E_1}$$
(Eq. 3)

$$\begin{array}{c} P \\ \hline e^{\uparrow} \\ \hline E_{1} \\ \hline E_{1} \\ \hline e^{\downarrow} b \\ \hline$$

(a) AXIAL FORCES

Solve simultaneously Eqs. (1) and (3):

.....

$$P_1 = \frac{PE_1}{E_1 + E_2} \quad P_2 = \frac{PE_2}{E_1 + E_2} \leftarrow$$

(b Eccentricity of load P

Substitute P_1 and P_2 into Eq. (2) and solve for e:

$$e = \frac{b(E_2 - E_1)}{2(E_2 + E_1)} \longleftarrow$$

(c) RATIO OF STRESSES

$$\sigma_1 \frac{P_1}{A} \quad \sigma_2 = \frac{P_2}{A} \quad \frac{\sigma_1}{\sigma_2} = \frac{P_1}{P_2} = \frac{E_1}{E_2} \longleftarrow$$

Problem 2.4-12 A circular steel bar *ABC* (E = 200 GPa) has crosssectional area A_1 from *A* to *B* and cross-sectional area A_2 from *B* to *C* (see figure). The bar is supported rigidly at end *A* and is subjected to a load *P* equal to 40 kN at end *C*. A circular steel collar *BD* having cross-sectional area A_3 supports the bar at *B*. The collar fits snugly at *B* and *D* when there is no load.

Determine the elongation δ_{AC} of the bar due to the load *P*. (Assume $L_1 = 2L_3 = 250 \text{ mm}$, $L_2 = 225 \text{ mm}$, $A_1 = 2A_3 = 960 \text{ mm}^2$, and $A_2 = 300 \text{ mm}^2$.)

.....

Solution 2.4-12 Bar supported by a collar

FREE-BODY DIAGRAM OF BAR ABC AND COLLAR BD

EQUILIBRIUM OF BAR ABC

$$\Sigma F_{\text{vert}} = 0 \qquad R_A + R_D - P = 0 \qquad (\text{Eq. 1})$$

COMPATIBILITY (distance AD does not change)

 $\delta_{AB}(\text{bar}) + \delta_{BD}(\text{collar}) = 0$ (Eq. 2) (Elongation is positive.)

FORCE-DISPLACEMENT RELATIONS

$$\delta_{AB} = \frac{R_A L_1}{EA_1} \qquad \delta_{BD} = -\frac{R_D L_3}{EA_3}$$

Substitute into Eq. (2):
$$\frac{R_A L_1}{EA_1} - \frac{R_D L_3}{EA_3} = 0 \qquad (Eq. 3)$$

Solve simultaneously Eqs. (1) and (3):

$$R_A = \frac{PL_3A_1}{LA_3 + L_3A_1} \quad R_D = \frac{PL_1A_3}{L_1A_3 + L_3A_1}$$

CHANGES IN LENGTHS (Elongation is positive)

$$\delta_{AB} = \frac{P_A L_1}{EA_1} = \frac{P L_1 L_3}{E(L_1 A_3 + L_3 A_1)} \qquad \delta_{BC} = \frac{P L_2}{EA_2}$$

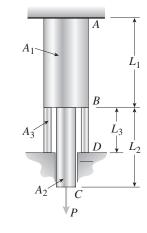
ELONGATION OF BAR ABC

$$\delta_{AC} = \delta_{AB} + \delta_{AC}$$

.....

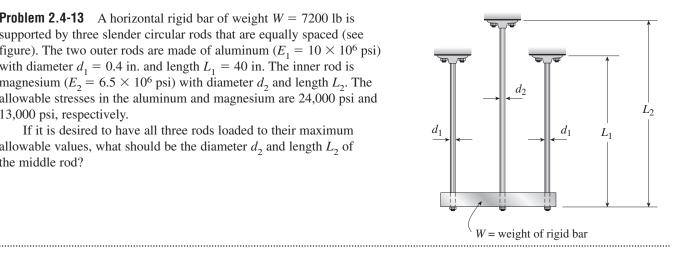
SUBSTITUTE NUMERICAL VALUES:

$$P = 40 \text{ kN}$$
 E = 200 GPa
L₁ = 250 mm
L₂ = 225 mm
L₃ = 125 mm
A₁ = 960 mm²
A₂ = 300 mm²
A₃ = 480 mm²
RESULTS:
R_A = R_D = 20 kN
δ_{AB} = 0.02604 mm
δ_{BC} = 0.15000 mm
δ_{AC} = δ_{AB} + δ_{AC} = 0.176 mm ←

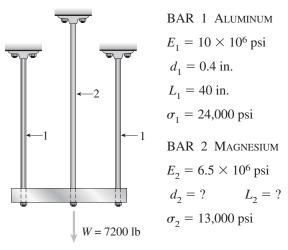


Problem 2.4-13 A horizontal rigid bar of weight W = 7200 lb is supported by three slender circular rods that are equally spaced (see figure). The two outer rods are made of aluminum ($E_1 = 10 \times 10^6 \text{ psi}$) with diameter $d_1 = 0.4$ in. and length $L_1 = 40$ in. The inner rod is magnesium ($E_2 = 6.5 \times 10^6$ psi) with diameter d_2 and length L_2 . The allowable stresses in the aluminum and magnesium are 24,000 psi and 13,000 psi, respectively.

If it is desired to have all three rods loaded to their maximum allowable values, what should be the diameter d_2 and length L_2 of the middle rod?



Solution 2.4-13 Bar supported by three rods



FREE-BODY DIAGRAM OF RIGID BAR EQUATION OF EQUILIBRIUM

$$\begin{array}{c}
 \uparrow F_1 \\
 \uparrow F_2 \\
 W \\
 \hline W
\end{array}$$

$$\begin{array}{c}
 \Sigma F_{vert} = 0 \\
 2F_1 + F_2 - W = 0 \quad (Eq. 1)
\end{array}$$

FULLY STRESSED RODS

$$F_{1} = \sigma_{1}A_{1} \qquad F_{2} = \sigma_{2}A_{2}$$
$$A_{1} = \frac{\pi d_{1}^{2}}{4} \qquad A_{2} = \frac{\pi d_{2}^{2}}{4}$$

Substitute into Eq. (1):

$$2\sigma_1\left(\frac{\pi d_1^2}{4}\right) + \sigma_2\left(\frac{\pi d_2^2}{4}\right) = W$$

Diameter d_1 is known; solve for d_2 :

$$d_2^2 = \frac{4W}{\pi\sigma_2} - \frac{2\sigma_1 d_2^2}{\sigma_2} \longleftarrow$$
(Eq. 2)

SUBSTITUTE NUMERICAL VALUES:

$$d_2^2 = \frac{4(7200 \text{ lb})}{\pi(13,000 \text{ psi})} - \frac{2(24,000 \text{ psi}) (0.4 \text{ in.})^2}{13,000 \text{ psi}}$$

= 0.70518 in.² - 0.59077 in.² = 0.11441 in.²
$$d_2 = 0.338 \text{ in.} \longleftarrow$$

EQUATION OF COMPATIBILITY

$$\delta_1 = \delta_2 \tag{Eq. 3}$$

FORCE-DISPLACEMENT RELATIONS

$$\delta_1 = \frac{F_1 L_1}{E_1 A_1} = \sigma_1 \left(\frac{L_1}{E_1}\right) \tag{Eq. 4}$$

$$\delta_2 = \frac{F_2 L_2}{E_2 A_2} = \sigma_2 \left(\frac{L_2}{E_2}\right) \tag{Eq. 5}$$

Substitute (4) and (5) into Eq. (3):

$$\sigma_1\left(\frac{L_1}{E_1}\right) = \sigma_2\left(\frac{L_2}{E_2}\right)$$

Length L_1 is known; solve for L_2 :

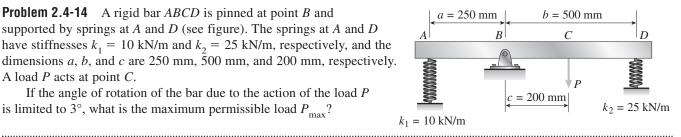
$$L_2 = L_1 \left(\frac{\sigma_1 E_2}{\sigma_2 E_1} \right) \longleftarrow$$
 (Eq. 6)

SUBSTITUTE NUMERICAL VALUES:

$$L_2 = (40 \text{ in.}) \left(\frac{24,000 \text{ psi}}{13,000 \text{ psi}}\right) \left(\frac{6.5 \times 10^6 \text{ psi}}{10 \times 10^6 \text{ psi}}\right)$$

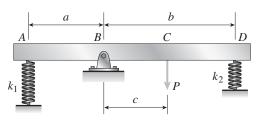
= 48.0 in.

Problem 2.4-14 A rigid bar *ABCD* is pinned at point *B* and supported by springs at A and D (see figure). The springs at A and D have stiffnesses $k_1 = 10$ kN/m and $k_2 = 25$ kN/m, respectively, and the dimensions a, b, and c are 250 mm, 500 mm, and 200 mm, respectively. A load P acts at point C.



If the angle of rotation of the bar due to the action of the load Pis limited to 3°, what is the maximum permissible load P_{max} ?

Solution 2.4-14 Rigid bar supported by springs



NUMERICAL DATA

$$a = 250 \text{ mm}$$

$$b = 500 \text{ mm}$$

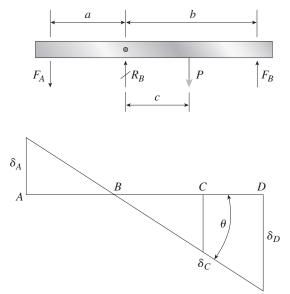
c = 200 mm

$$k_1 = 10 \text{ kN/m}$$

$$k_2 = 25 \text{ kN/m}$$

$$\theta_{\rm max} = 3^\circ = \frac{\pi}{60}$$
 rad

FREE-BODY DIAGRAM AND DISPLACEMENT DIAGRAM



EQUATION OF EQUILIBRIUM

$$\Sigma M_B = 0 \quad \text{for } F_A(a) - P(c) + F_D(b) = 0 \qquad \text{(Eq. 1)}$$

EQUATION OF COMPATIBILITY

å

$$\frac{\delta_A}{a} = \frac{\delta_D}{b} \tag{Eq. 2}$$

FORCE-DISPLACEMENT RELATIONS

$$\delta_A = \frac{F_A}{k_1} \qquad \delta_D = \frac{F_D}{k_2} \tag{Eqs. 3, 4}$$

SOLUTION OF EQUATIONS

Substitute (3) and (4) into Eq. (2):

$$\frac{F_A}{ak_1} = \frac{F_D}{bk_2} \tag{Eq. 5}$$

Solve simultaneously Eqs. (1) and (5):

$$F_A = \frac{ack_1P}{a^2k_1 + b^2k_2} \quad F_D = \frac{bck_2P}{a^2k_1 + b^2k_2}$$

ANGLE OF ROTATION

$$\delta_D = \frac{F_D}{k_2} = \frac{bcP}{a^2k_1 + b^2k_2} \quad \theta = \frac{\delta_D}{b} = \frac{cP}{a^2k_1 + b^2k_2}$$

MAXIMUM LOAD

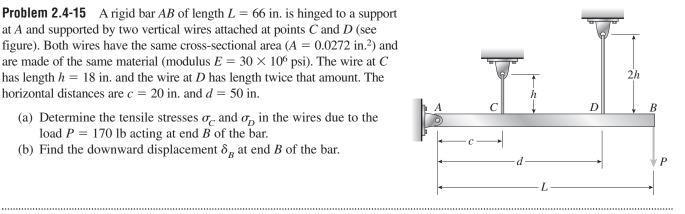
$$P = \frac{\theta}{c} (a^2 k_1 + b^2 k_2)$$
$$P_{\text{max}} = \frac{\theta_{\text{max}}}{c} (a^2 k_1 + b^2 k_2) \leftarrow$$

SUBSTITUTE NUMERICAL VALUES:

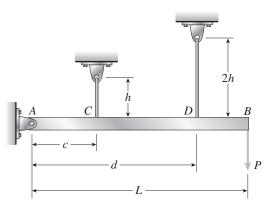
$$P_{\text{max}} = \frac{\pi/60 \text{ rad}}{200 \text{ mm}} [(250 \text{ mm})^2 (10 \text{ kN/m}) + (500 \text{ mm})^2 (25 \text{ kN/m})]$$
$$= 1800 \text{ N} \longleftarrow$$

Problem 2.4-15 A rigid bar *AB* of length L = 66 in. is hinged to a support at A and supported by two vertical wires attached at points C and D (see figure). Both wires have the same cross-sectional area (A = 0.0272 in.²) and are made of the same material (modulus $E = 30 \times 10^6$ psi). The wire at C has length h = 18 in. and the wire at D has length twice that amount. The horizontal distances are c = 20 in. and d = 50 in.

- (a) Determine the tensile stresses $\sigma_{\!C}$ and $\sigma_{\!D}$ in the wires due to the load P = 170 lb acting at end B of the bar.
- (b) Find the downward displacement δ_B at end *B* of the bar.







h = 18 in.

2h = 36 in.

c = 20 in.

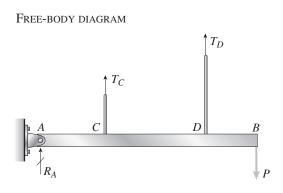
$$d = 50 \text{ in}.$$

L = 66 in.

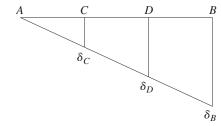
 $E = 30 \times 10^6 \text{ psi}$

 $A = 0.0272 \text{ in.}^2$

 $P = 340 \, \text{lb}$







EQUATION OF EQUILIBRIUM

$$\Sigma M_A = 0 \quad \text{(Eq. 1)} \quad T_c(c) + T_D(d) = PL \quad (\text{Eq. 1})$$

EQUATION OF COMPATIBILITY

$$\frac{\delta_c}{c} = \frac{\delta_d}{d} \tag{Eq. 2}$$

FORCE-DISPLACEMENT RELATIONS

$$\delta_c = \frac{T_c h}{EA}$$
 $\delta_D = \frac{T_D(2h)}{EA}$ (Eqs. 3, 4)

SOLUTION OF EQUATIONS

Substitute (3) and (4) into Eq. (2):

$$\frac{T_c h}{cEA} = \frac{T_D(2h)}{dEA} \quad \text{or} \quad \frac{T_c}{c} = \frac{2T_D}{d}$$
(Eq. 5)

TENSILE FORCES IN THE WIRES

Solve simultaneously Eqs. (1) and (5):

$$T_c = \frac{2cPL}{2c^2 + d^2}$$
 $T_D = \frac{dPL}{2c^2 + d^2}$ (Eqs. 6, 7)

TENSILE STRESSES IN THE WIRES

$$\sigma_c = \frac{T_c}{A} = \frac{2cPL}{A(2c^2 + d^2)}$$
 (Eq. 8)

$$\sigma_D = \frac{T_D}{A} = \frac{dPL}{A(2c^2 + d^2)}$$
(Eq. 9)

DISPLACEMENT AT END OF BAR

$$\delta_B = \delta_D \left(\frac{L}{d}\right) = \frac{2hT_D}{EA} \left(\frac{L}{d}\right) = \frac{2hPL^2}{EA(2c^2 + d^2)} \quad \text{(Eq. 10)}$$

SUBSTITUTE NUMERICAL VALUES

$$2c^{2} + d^{2} = 2(20 \text{ in.})^{2} + (50 \text{ in.})^{2} = 3300 \text{ in.}^{2}$$
(a) $\sigma_{c} = \frac{2cPL}{A(2c^{2} + d^{2})} = \frac{2(20 \text{ in.})(340 \text{ lb})(66 \text{ in.})}{(0.0272 \text{ in.}^{2})(3300 \text{ in.}^{2})}$

$$= 10,000 \text{ psi} \longleftarrow$$

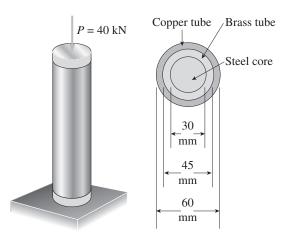
$$\sigma_D = \frac{dPL}{A(2c^2 + d^2)} = \frac{(50 \text{ in.})(340 \text{ lb})(66 \text{ in.})}{(0.0272 \text{ in.}^2)(3300 \text{ in.}^2)}$$
$$= 12,500 \text{ psi} \longleftarrow$$

(b)
$$\delta_B = \frac{2hPL^2}{EA(2c^2 + d^2)}$$

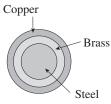
$$= \frac{2(18 \text{ in.})(340 \text{ lb})(66 \text{ in.})^2}{(30 \times 10^6 \text{ psi})(0.0272 \text{ in.}^2)(3300 \text{ in.}^2)}$$
$$= 0.0198 \text{ in.} \longleftarrow$$

Problem 2.4-16 A trimetallic bar is uniformly compressed by an axial force P = 40 kN applied through a rigid end plate (see figure). The bar consists of a circular steel core surrounded by brass and copper tubes. The steel core has diameter 30 mm, the brass tube has outer diameter 45 mm, and the copper tube has outer diameter 60 mm. The corresponding moduli of elasticity are $E_s = 210$ GPa, $E_b = 100$ GPa, and $E_c = 120$ GPa.

Calculate the compressive stresses σ_s , σ_b , and σ_c in the steel, brass, and copper, respectively, due to the force *P*.



Solution 2.4-16 Trimetallic bar in compression



 P_s = compressive force in steel core

 $P_b =$ compressive force in brass tube

 $P_c =$ compressive force in copper tube

FREE-BODY DIAGRAM OF RIGID END PLATE



EQUATION OF EQUILIBRIUM

$$\Sigma F_{\text{vert}} = 0$$
 $P_s + P_b + P_c = P$ (Eq. 1)

EQUATIONS OF COMPATIBILITY

$$\delta_s = \delta_B \qquad \delta_c = \delta_s \qquad (Eqs. 2)$$

FORCE-DISPLACEMENT RELATIONS

$$\delta_s = \frac{P_s L}{E_s A_s} \quad \delta_b = \frac{P_b L}{E_b A_b} \quad \delta_c = \frac{P_c L}{E_c A_c} \quad (\text{Eqs. 3, 4, 5})$$

SOLUTION OF EQUATIONS

Substitute (3), (4), and (5) into Eqs. (2):

$$P_b = P_s \frac{E_b A_b}{E_s A_s} \quad P_c = P_s \frac{E_c A_c}{E_s A_s}$$
(Eqs. 6, 7)

Solve simultaneously Eqs. (1), (6), and (7):

$$P_{s} = P \frac{E_{s} A_{s}}{E_{s} A_{s} + E_{b} A_{b} + E_{c} A_{c}}$$
$$P_{b} = P \frac{E_{b} A_{b}}{E_{s} A_{s} + E_{b} A_{b} + E_{c} A_{c}}$$
$$P_{c} = P \frac{E_{c} A_{c}}{E_{s} A_{s} + E_{b} A_{b} + E_{c} + A_{c}}$$

COMPRESSIVE STRESSES

Let
$$\Sigma EA = E_s A_s + E_b A_b + E_c A_c$$

 $\sigma_s = \frac{P_s}{A_s} = \frac{P E_s}{\Sigma EA}$ $\sigma_b = \frac{P_b}{A_b} = \frac{P E_b}{\Sigma EA}$
 $\sigma_c = \frac{P_c}{A_c} = \frac{P E_c}{\Sigma EA}$

 $SUBSTITUTE \ NUMERICAL \ VALUES:$

$$P = 40 \text{ kN} \qquad E_s = 210 \text{ GPa}$$

$$E_b = 100 \text{ GPa} \qquad E_c = 120 \text{ GPa}$$

$$d_1 = 30 \text{ mm} \qquad d_2 = 45 \text{ mm} \qquad d_3 = 60 \text{ mm}$$

$$A_s = \frac{\pi}{4} d_1^2 = 706.86 \text{ mm}^2$$

$$A_b = \frac{\pi}{4} (d_2^2 - d_1^2) = 883.57 \text{ mm}^2$$

$$A_c = \frac{\pi}{4} (d_3^2 - d_2^2) = 1237.00 \text{ mm}^2$$

$$\Sigma EA = 385.238 \times 10^6 \text{ N}$$

$$\sigma_s = \frac{PE_s}{\Sigma EA} = 21.8 \text{ MPa} \longleftarrow$$

$$\sigma_b = \frac{PE_b}{\Sigma EA} = 10.4 \text{ MPa} \longleftarrow$$

$$\sigma_c = \frac{PE_c}{\Sigma EA} = 12.5 \text{ MPa} \longleftarrow$$

Thermal Effects

Problem 2.5-1 The rails of a railroad track are welded together at their ends (to form continuous rails and thus eliminate the clacking sound of the wheels) when the temperature is 60° F.

What compressive stress σ is produced in the rails when they are heated by the sun to 120°F if the coefficient of thermal expansion $\alpha = 6.5 \times 10^{-6}$ /°F and the modulus of elasticity $E = 30 \times 10^{6}$ psi?

Solution 2.5-1 Expansion of railroad rails

The rails are prevented from expanding because of	$\Delta T = 120^{\circ}\mathrm{F} - 60^{\circ}\mathrm{F} = 60^{\circ}\mathrm{F}$
their great length and lack of expansion joints.	$\sigma = E\alpha(\Delta T)$
Therefore, each rail is in the same condition as a bar with fixed ends (see Example 2-7).	= $(30 \times 10^6 \text{ psi})(6.5 \times 10^{-6})(60^\circ \text{F})$
The compressive stress in the rails may be calculated from Eq. (2-18).	$\sigma = 11,700 \text{ psi} \longleftarrow$

Problem 2.5-2 An aluminum pipe has a length of 60 m at a temperature of 10°C. An adjacent steel pipe at the same temperature is 5 mm longer than the aluminum pipe.

At what temperature (degrees Celsius) will the aluminum pipe be 15 mm longer than the steel pipe? (Assume that the coefficients of thermal expansion of aluminum and steel are $\alpha_a = 23 \times 10^{-6}$ /°C and $\alpha_s = 12 \times 10^{-6}$ /°C, respectively.)

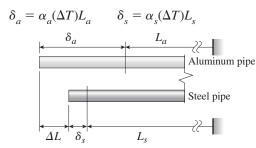
Solution 2.5-2 Aluminum and steel pipes

INITIAL CONDITIONS	
$L_a = 60 \text{ m}$	$T_0 = 10^{\circ} \text{C}$
$L_s = 60.005 \text{ m}$	$T_0 = 10^{\circ} \text{C}$
$\alpha_a = 23 \times 10^{-6} / ^{\circ} \mathrm{C}$	$\alpha_s = 12 \times 10^{-6} / ^{\circ}\mathrm{C}$

FINAL CONDITIONS

Aluminum pipe is longer than the steel pipe by the amount $\Delta L = 15$ mm.

 ΔT = increase in temperature



From the figure above:

$$\delta_a + L_a = \Delta L + \delta_s + L_s$$

or,
$$\alpha_a(\Delta T)L_a + L_a = \Delta L + \alpha_s(\Delta T)L_s + L_s$$

Solve for ΔT :

Solve for ΔI :

$$\Delta T = \frac{\Delta L + (L_s - L_a)}{\alpha_a L_a - \alpha_s L_s}$$

Substitute numerical values:

$$\alpha_a L_a - \alpha_s L_s = 659.9 \times 10^{-6} \text{ m/°C}$$
$$\Delta T = \frac{15 \text{ mm} + 5 \text{ mm}}{659.9 \times 10^{-6} \text{ m/°C}} = 30.31^{\circ}\text{C}$$
$$T = T_0 + \Delta T = 10^{\circ}\text{C} + 30.31^{\circ}\text{C}$$
$$= 40.3^{\circ}\text{C} \quad \longleftarrow$$